

Acoustic Interests and Activities of the US Navy

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SERDP • ESTCP
SYMPOSIUM

2018 | Enhancing DoD's Mission Effectiveness

Outline

- Navy Environment
- The Cost of Hearing Loss
- Commercial vs Tactical Aircraft Engines
- Where Does Noise Come From?
- How to Deal With it
- What we are Doing
- Challenges
- Navy Capabilities
- Funded Programs
- Summary

Why am I Here?



The Cost of Hearing Loss

- Tinnitus and Hearing Most Prevalent Disabilities
- Hard to determine cost – estimates over a billion a year
- Additional Impacts to Operations/Communications etc...

Quick Reference: Recipients, Payments and Disabilities

Totals – new recipients

Veterans who began receiving compensation benefits	313,052
Survivors who began receiving service-connected death benefits	28,029
Total:	341,081

Estimated annual amounts paid – new recipients

Benefit program	Number of recipients	Estimated average individual amount paid annually	Estimated total amount paid annually
Compensation	313,052	\$11,558	\$3.62 Billion
Service-Connected Death ¹	28,029	\$15,961	\$447 Million
Total:	341,081	\$11,920	\$4.07 Billion

Most prevalent service-connected (SC) disabilities of new compensation recipients

Disability	Number of recipients
Tinnitus	157,848
Hearing loss	85,327
Lumbosacral or cervical strain	80,748
Limitation of flexion, knee	80,353
Scars, general	64,788
Post-traumatic stress disorder	63,049
Limitation of motion of the ankle	51,622
Migraine	45,840
Impairment of the knee, general	37,395
Bursitis	36,194
Total number of most prevalent disabilities	703,164
Total number of disabilities²	1,642,994

Average number of SC disabilities per new compensation recipient

New Recipients: 313,052	Total number of disabilities: 1,642,994
Average SC disabilities per Veteran: 5.25	

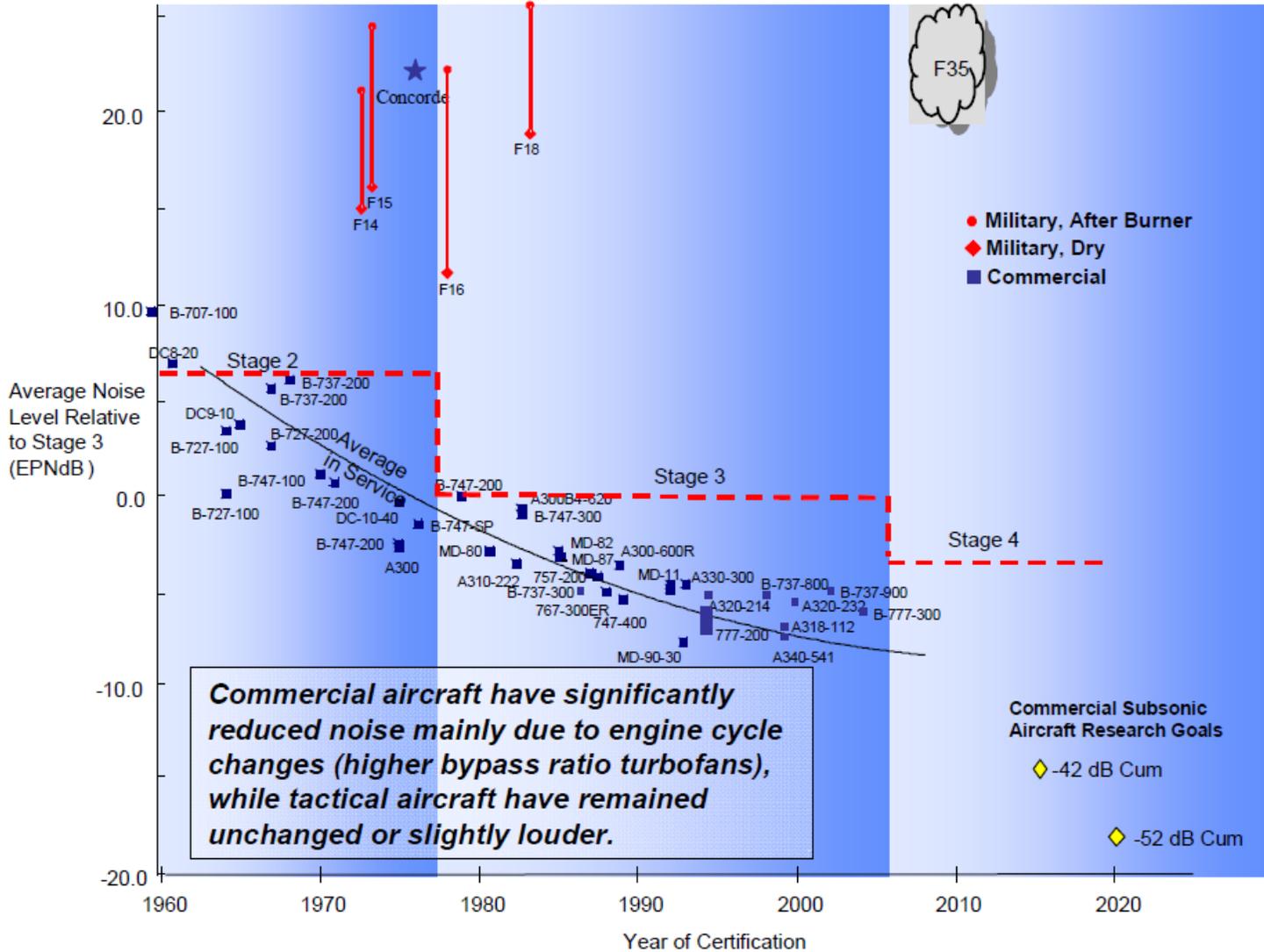
U.S. Department of Veterans Affairs
Veterans Benefits Administration

¹ Dependency and Indemnity Compensation and Death Compensation.
² See page 25 for more information.

VA

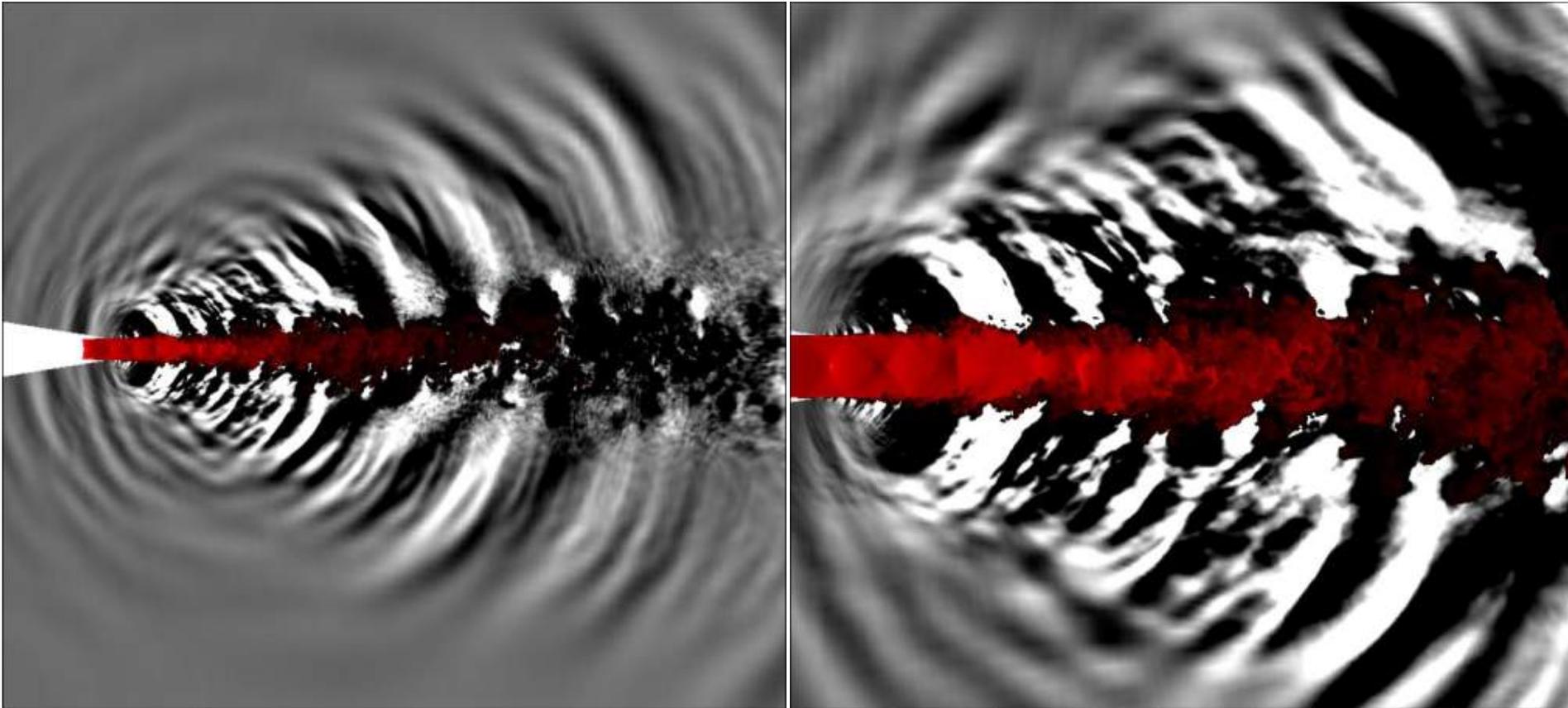
Compensation – Page 4 of 52

Commercial vs Tactical Aircraft Engines



Where Does Jet Noise Come From?

Visualization of sound waves radiated from an over-expanded hot supersonic jet;
 $Tr=1.74, M_{jet}= 1.35$



How to Deal With It

- Source
 - Reduce Jet Velocity
 - Enhance Jet Mixing (like Chevrons)
 - Other Methods Show Promise in Laboratories, but need Further Development
- Path
 - Hearing Protection
 - Acoustic Enclosures/Barriers
- Operations
 - Minimize Exposure Time
 - Noise Abatement Procedures

How to Deal With It



Chevrons
In Service On Commercial Engines
Struggling to Transition

What We Are Doing

- USN Has a Noise Induced Hearing Loss Program
- Hearing Protection and Communications are being worked
- ONR Code 35 Has an S&T Jet Noise Reduction Program
 - Consistent Funding
 - Working Simulations (Prediction), Diagnostics and Noise Reduction Technologies
 - Collaborating with AF and NASA where possible
 - Full Scale Measurements, Diagnostics, Simulations

Challenges

- No Requirement or KPP
 - Noise is not considered in design/development
- Unable to trade capability for noise
 - Much harder to apply a band-aid
- Hard Problem – Physics are Physics
 - These are the highest performance aircraft
- Commercial Engines Have Designed Jet Noise out of the picture
 - High Bypass Ratio Lowers the Exhaust Velocity and is good for fuel efficiency
 - Commercial Noise Certification Requirements Continue to Become More Stringent
- \$\$

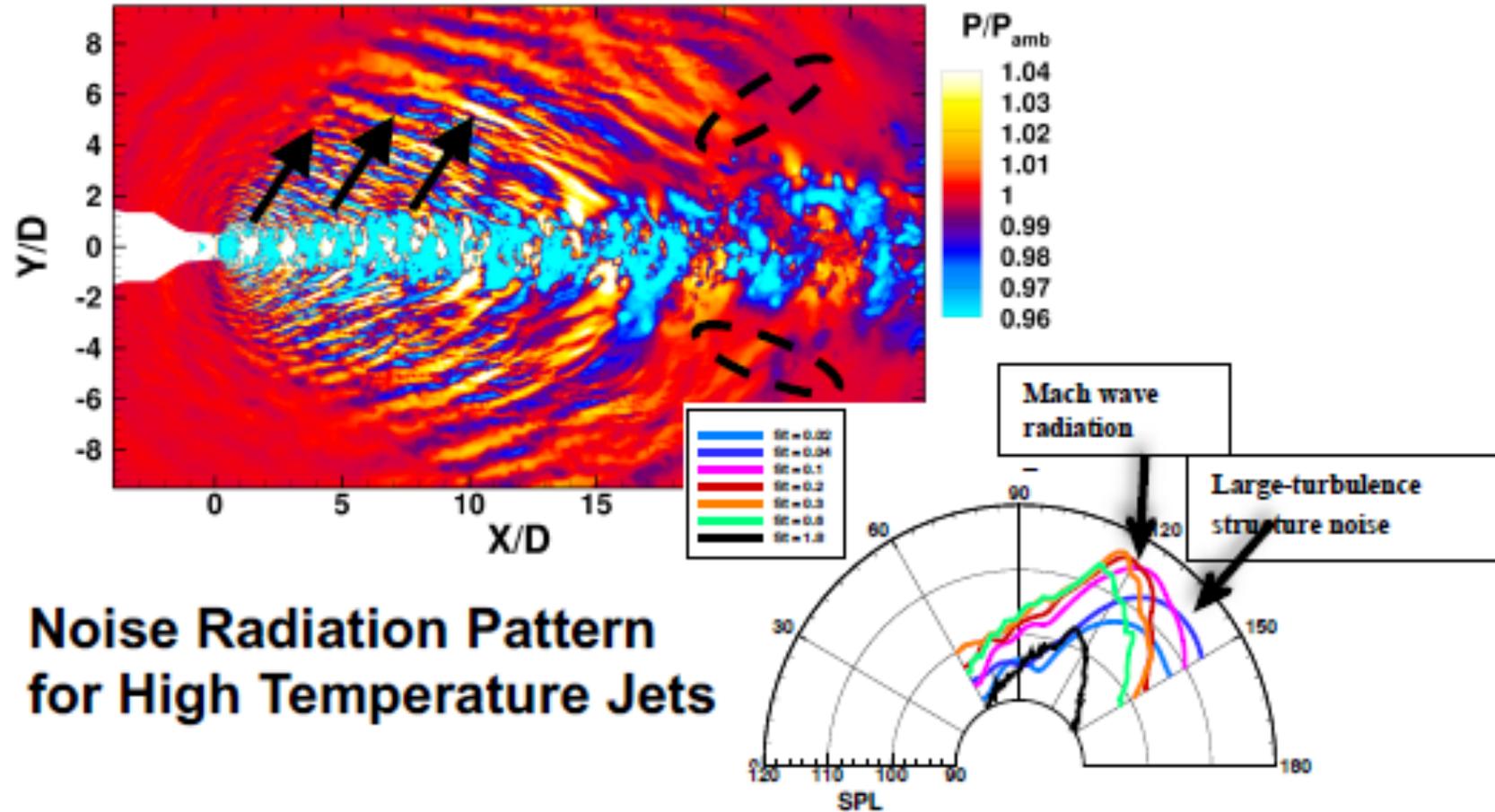
Navy Capabilities – NRL – High Fidelity Simulations

Objective/Description

- Develop , validate and apply a high-fidelity, modeling and simulation tool (called JENRE) for jet noise characterization, prediction, reduction and optimization.

Key Technologies

- Fundamental insight into sources and directivity of jet noise during Naval operations.
- Validated prediction tool for future low-noise designs.



Navy Capabilities – NAVAIR – Full Scale Measurements and Data Analysis

Objective/Description

- Develop improved test planning guidelines / procedures / best practices for outdoor, full scale supersonic military jet noise measurements.
- Review full scale jet noise data taken on F/A-18E and F-35 and contrast and compare results.
- Develop feedback for ANSI S12.75 jet noise measurement standard committee for update of standard procedures.

Key Technologies

- Outdoor jet noise measurement equipment and procedures leading to accurate, reliable, repeatable outdoor noise measurement results



NAVAIR Noise Measurements - Proving the Difference

- In 2012 ONR/NAVAIR and F/A-18 program office funded development of 1st intentional noise reduction treatment for supersonic military aircraft: Chevrons.
 - Aircraft testing is expensive! (~\$30K/hour + + + +)
- ProNTO supported noise measurements under internal funding using current best practices.
- Challenge – prove Chevrons make a difference in back-to-back experiments.



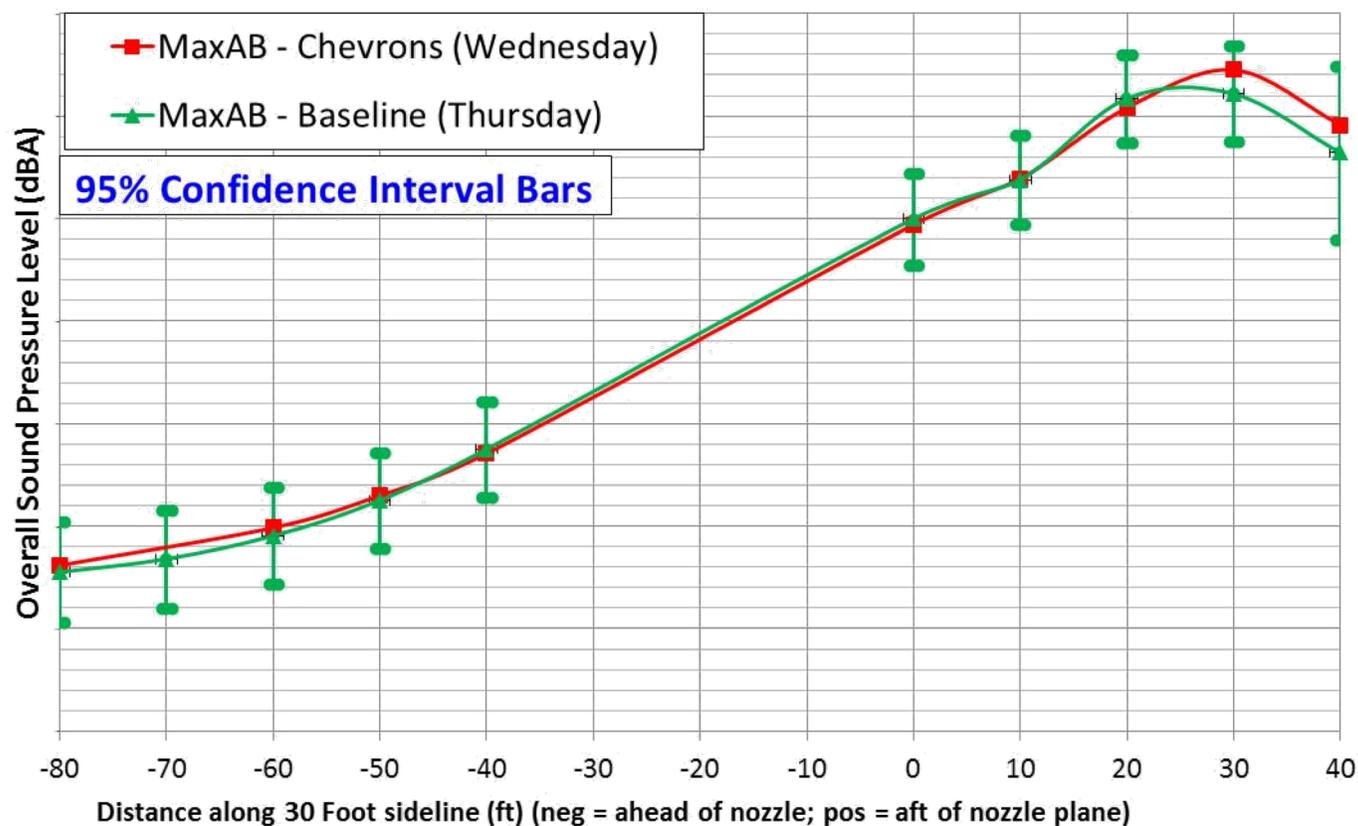
NAVAIR Noise Measurements - Proving the Difference

NAVAIR Noise Team

Results of Dec 18-20, 2012 Measurements of F-18E

With and Without Chevrons

3 foot High Mics, Line offset 30 Feet from AC Centerline



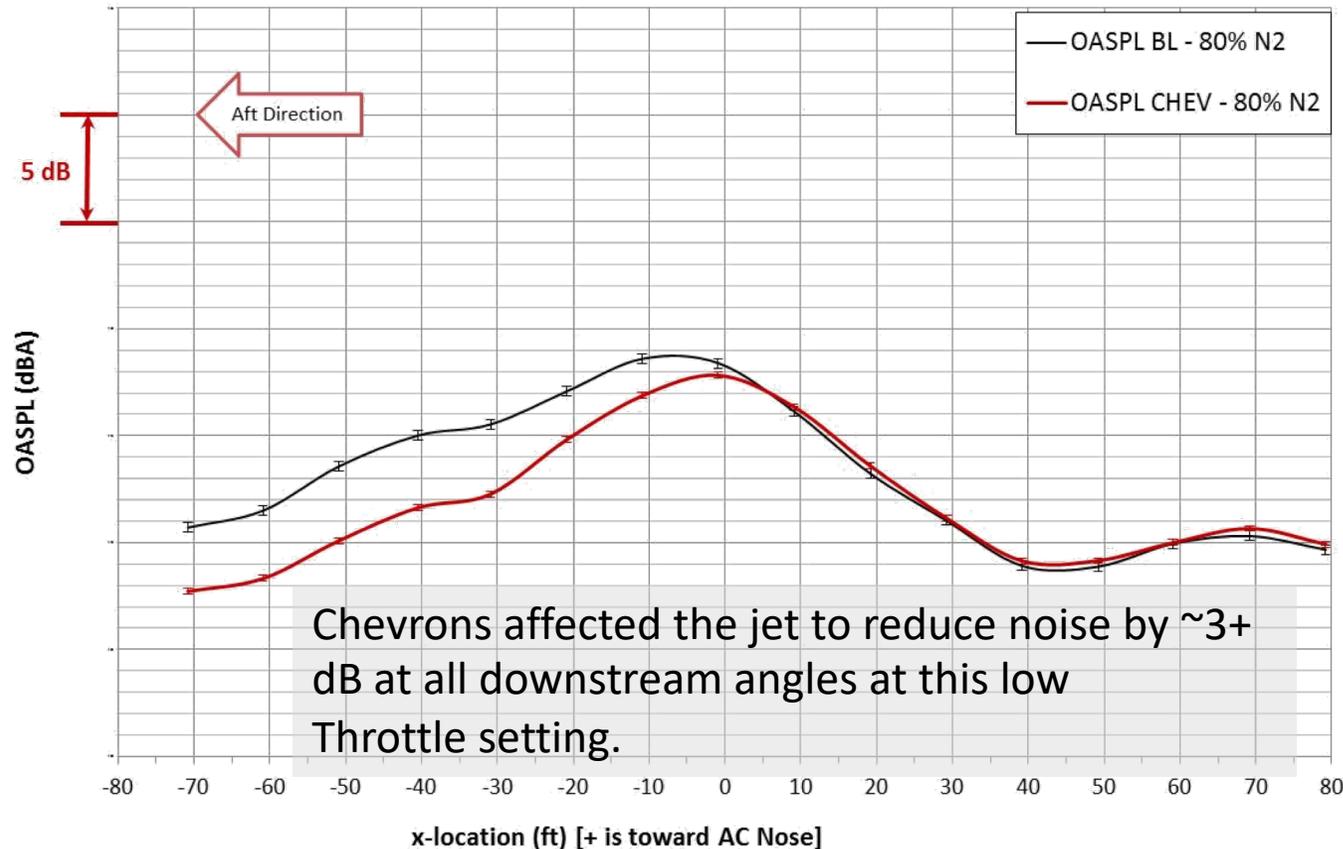
- We estimated 95% confidence interval: $\pm 2.8\text{dB}$ at the time – it's actually higher
- **ANSI S12.75 standard interval is $\pm 5.8\text{dB}$**
- Any difference within range is **statistically insignificant** when following minimum recommendations of ANSI S12.75-2012

NAVAIR Noise Measurements - Proving the Difference

- Program office not happy with that answer!
- Even if actually was a 3dB improvement – would not have been able to resolve it with those test methods – **even though they were industry best practices!** NOT GOOD ENOUGH!
- Dug into ANSI S12.75 Annex A and evaluated all the assumptions we had made – test plan, test limits
- Key contributors to uncertainty in outdoor supersonic aircraft noise measurements:
 1. **Wind** – causes noise source location to move – changes propagation pattern – different distortion test to test – (in 2012 moved from West to East between test days)
 2. **Test Aircraft throttle setting is not scientifically repeatable** – pilot difference, engine condition, FADEC, engine warmup
 3. **Number of repeats**
 4. Temperature gradients from sun, etc.

NAVAIR Noise Measurements – Chevrons Noise Reduction

F/A-18E Chevrons Noise Reduction Test, Oct 2014
30 Foot Line Data, 3 foot high mics, Left Engine Only @ 80% N2 Power



- **2014 Program Office made new plans to retest**
- Much more careful sculpting of test plan & test limits. PMA chose Lakehurst (LKE)
 - Ability to test at night
 - Best wind
 - No temperature gradient
 - Low background noise & EMI
- Planned for max repeats within fuel budget –
- Went to single engine + careful recording of engine parameters, corrected for thrust using cycle deck
- **RESULT: Answers ± 0.22 dB 95% Confidence Interval between measurements**

ONR – Funded Programs

- Krish Ahuja – Ga Tech ‘Development of Noise Prediction Design Tools for Future Multi-Stream, Low-Noise Tactical Aircraft (TACAIR) Engine Exhaust.’
- Todd Lowe – Va Tech ‘Turbulence Development in Non-Uniform, Heated Supersonic Jets for Noise Reduction’
- Tim Colonius – Cal Tech ‘Next Generation Jet Noise Models For Complex Geometry Nozzles’
- Steve Miller – U of Florida ‘Turbulent Statistics and Associated Acoustic Sources Near Shock Wave Shear Layer Interactions’
- Datta Gaitonde – OSU ‘Noise Generation Mechanisms in Imperfectly Expanded & High Speed Jets’
- Tony Pilon – LM ‘Acoustical Assessment of the Spiritech STAR Nozzle Concept

ONR – Funded Programs

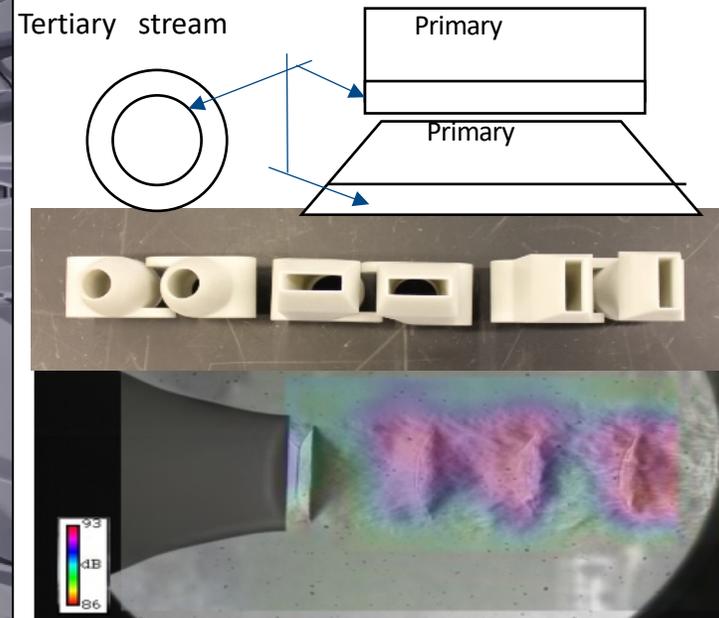
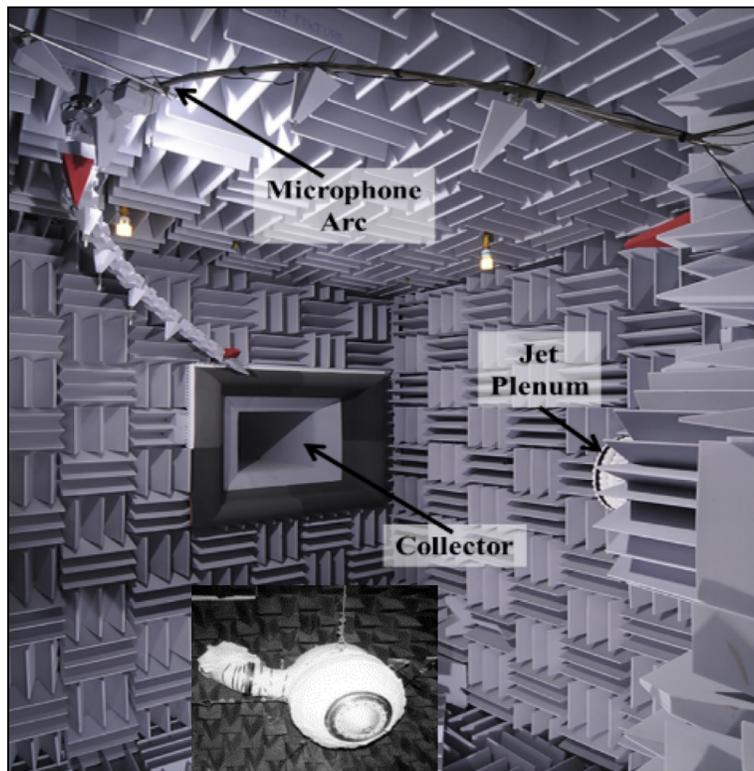
Krish Ahuja – Ga Tech ‘Development of Noise Prediction Design Tools for Future Multi-Stream, Low-Noise Tactical Aircraft (TACAIR) Engine Exhaust.’

Objective/Description

Acquire a relevant jet-noise database for coannular and non-coannular multi-stream jets operated at representative tactical aircraft exhaust conditions¹ to develop a semi-empirical noise prediction tool for representative carrier deck locations and in the far-field, for static jets² under the Base Phase

Payoffs/Key Technologies

See items 1 and 2 above



ONR – Funded Programs

Todd Lowe & Wing Ng– Va Tech ‘Turbulence Development in Non-Uniform, Heated Supersonic Jets for Noise Reduction’

Objective/Description

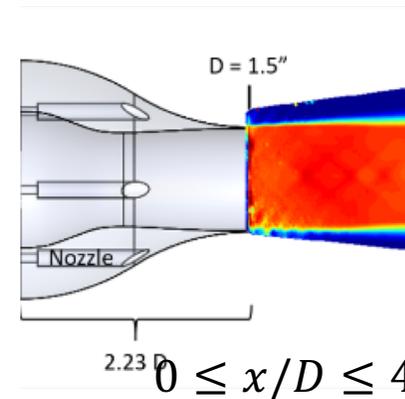
Experimental study of space/time plume turbulence characteristics for interpretation of full-scale results

Payoffs/Key Technologies

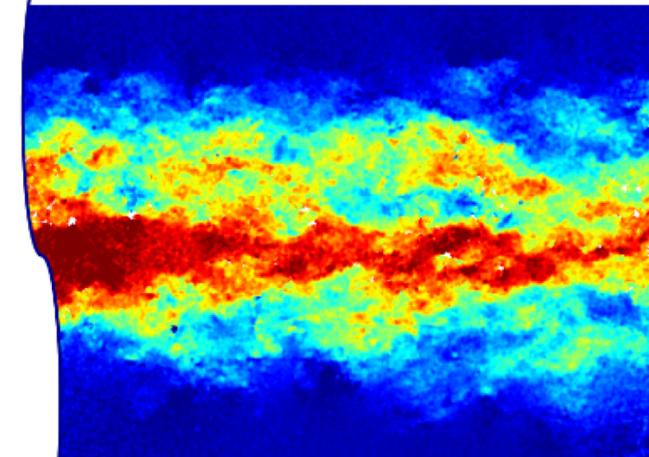
Plume turbulence info for noise prediction and reduction, computational validation

Scalable flow diagnostics

Near-nozzle mean U



Instantaneous U



Measurements can relate nozzle exit profiles to turbulence development.

ONR – Funded Programs

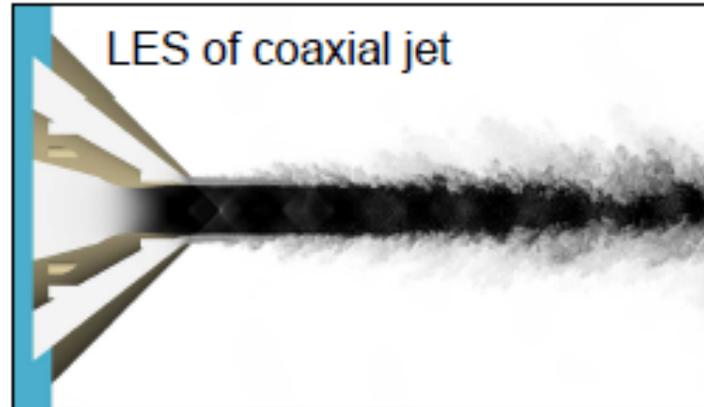
Tim Colonius – Cal Tech ‘Next Generation Jet Noise Models For Complex Geometry Nozzles’

Objective/Description

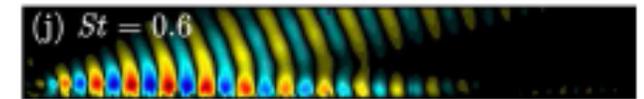
- Development of accurate, next-generation reduced-order models for the mixing noise of supersonic jets issuing from complex nozzle geometries

Key Technologies

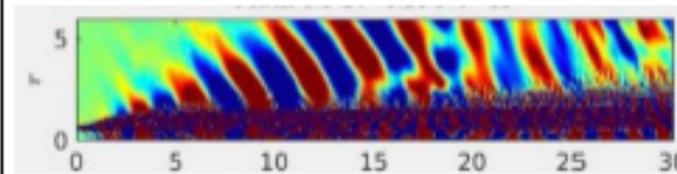
- Stochastically forced fast solvers for turbulence structure and radiated sound.
- Advanced methods for fast, spatial integration of the linearized Navier-Stokes equations (**O**ne-**W**ay **N**avier-**S**tokes equations, OWNS)



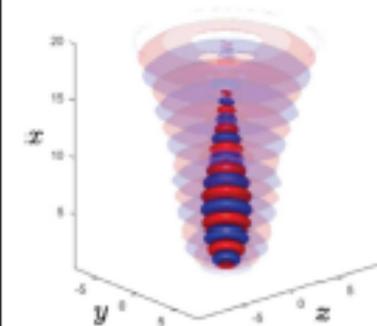
Resolvent analysis of axisymmetric jets



Forced and optimal OWNS



3D OWNS



ONR – Funded Programs

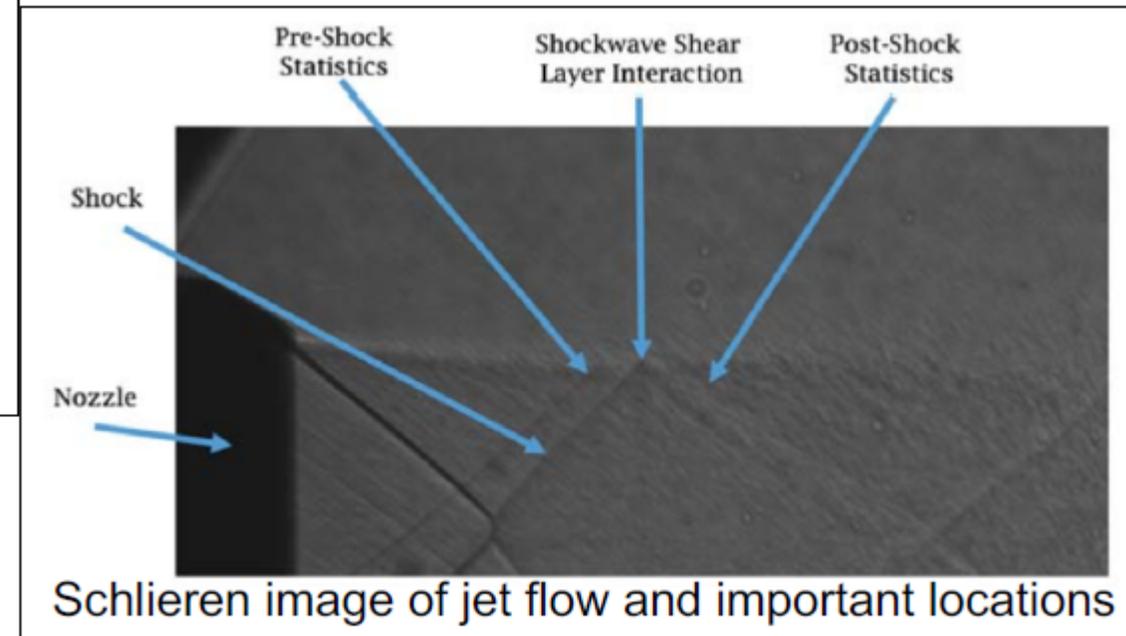
Steve Miller – U of Florida ‘Turbulent Statistics and Associated Acoustic Sources Near Shock Wave Shear Layer Interactions’

Objective/Description

- Provide guidance and physical understanding to reduce near-field noise from US Navy aircraft high speed jet exhaust
- Understand how turbulence statistics, wavenumber energy spectra, and the acoustic sources are changed by the shock cell structure
- Alter operating conditions to ascertain the effect on the statistics of turbulence and radiated noise
- Analytical model for the anisotropic portion of the turbulent field and connect the entire aeroacoustic model to large eddy simulation

Pay Offs / Key Technologies

- Combined analytical acoustic source modeling with large eddy simulation for understanding source of near-field jet noise. Guidance on methods to reduce shock-noise through alteration of source statistics. Reduction of hearing loss and sonic fatigue.



ONR – Funded Programs

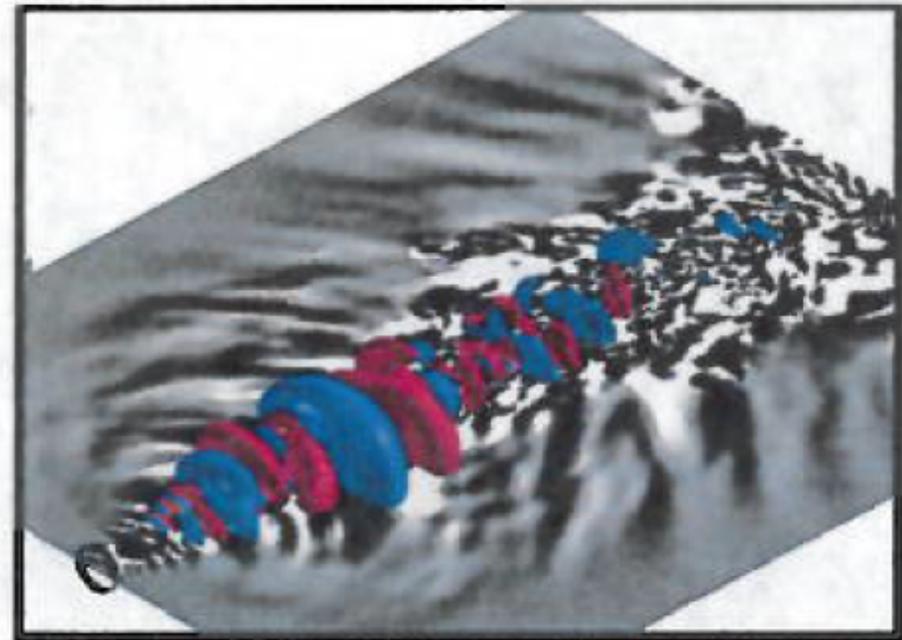
Datta Gaitonde – OSU ‘Noise Generation Mechanisms in Imperfectly Expanded & High Speed Jets’

Objective/Description

- Identify kinematic and thermodynamic acoustic energy sources in broad-band shock associated noise
- Noise source localization in overexpanded jets
- Acoustic receptivity to small perturbations

Key Technologies

- Time local linear tendency through synchronized Large-Eddy Simulations
- Momentum potential theory-based identification of acoustic energy sources



Acoustic mode and nearfield noise

ONR – Funded Programs

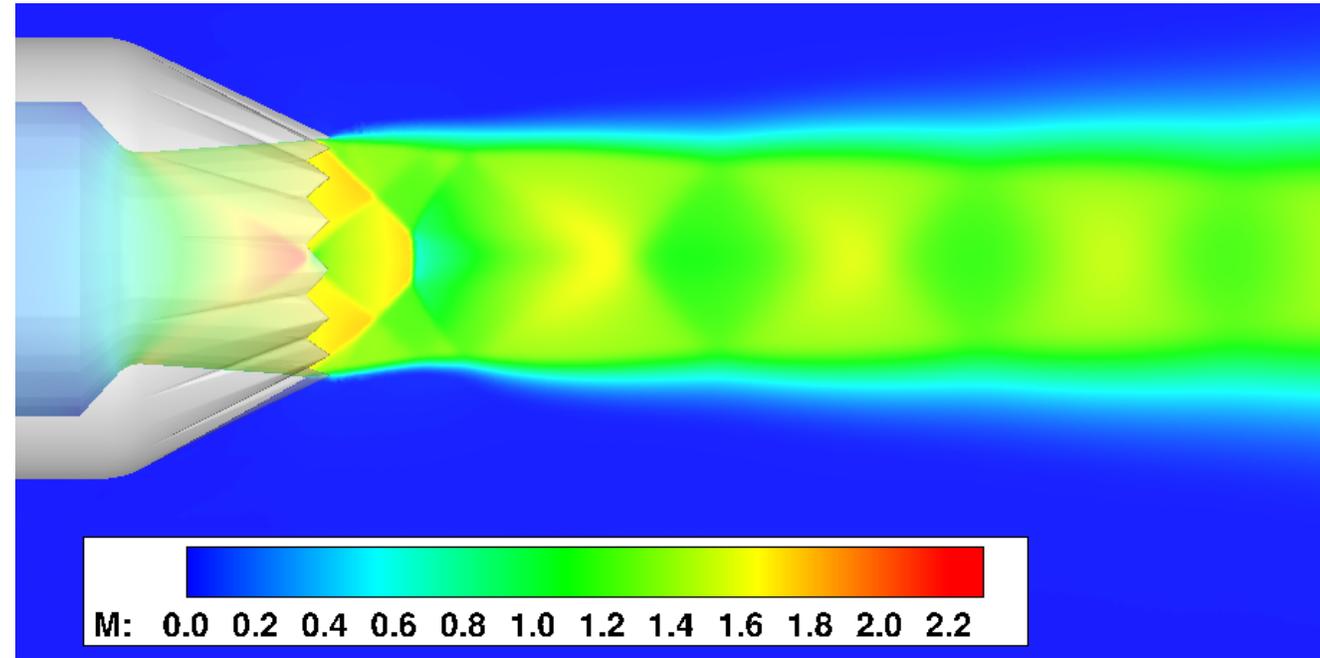
Tony Pilon – LM ‘Acoustical Assessment of the Spiritech STAR Nozzle Concept

Objective/Description

- Develop a STAR (Multi-Stream Coupled Throat Variable Rotating) Nozzle for integration with contemporary tactical aircraft employing advanced variable cycle engines
 - Nozzle design & cycle optimized for fuel efficiency and thrust
- Assess the acoustical properties and noise reduction potential of the STAR nozzle with respect to baseline nozzles employed on current aircraft
 - RANS (CFD++) and LES (JENRE) CFD analysis
 - Small-scale experiments in Penn State’s Jet Aeroacoustics facility

Payoffs/Key Technologies

- Low-noise, high efficiency nozzles for contemporary and next generation tactical aircraft
- Possible significant reductions in flight deck noise levels
- Industrial employment and assessment of NRL-developed *JENRE* tool



RANS Mach Contours – takeoff power, baseline nozzle

ONR – Funded Programs

- Jet Noise Remains a Safety and Good Neighbor Issue
- **Looking for innovative Jet noise reduction ideas**
- High Fidelity Simulations are Accurate But Computationally Expensive
- Looking for Opportunities for Full Scale Diagnostic Testing
- Challenging to Transition

Questions?

